On the stability and accuracy of high stiffness rendering in non-backdrivable actuators through series elasticity

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Abstract

This paper addresses the problem of accuracy and coupled stability of stiffnesscontrolled series elastic actuators, where the motor is modeled as a non-backdrivable velocity source, and the desired value of virtual stiffness is above the physical stiffness of the compliant element. We first demonstrate that, within the mentioned conditions, no linear outer-loop force control action can be applied on the velocity-sourced motor to passify the system. Relaxing the constraint of passivity, we exhaustively search the control design space defined by parametric force and stiffness controllers, expressed in a general lead-lag form, and define a lead-type stiffness compensator that results in acceptable conditions for both coupled stability and accuracy. We also address the effect of a non-ideality in the velocity control loop, such as limited-bandwidth velocity control, and derive relationships between the value of the inner velocity loop time constant and parameters of the stiffness compensator that provide the best performance in terms of both stability and accuracy of haptic display, and test our optimized controller both through numerical simulations and through experiments with the MR-SEA II.

We show that the parameters of a simple outer-loop stiffness compensator can be optimized to result in a stable and accurate display of virtual environments with stiffness values in a large range, that also comprises values of virtual stiffness *higher* than the physical stiffness of the compliant element. A

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requirement for coupled stability is that the actuator is designed such that the minimum value of inertia connected to the compliant actuator load is higher than a control-defined threshold. Finally, we extensively analyze how the minimum value of interaction mass for coupled stability can be minimized through modulation of the stiffness compensator zeros and poles, considering realistic limitations in the velocity control bandwidth of non-backdrivable motors. Our analysis, validated through both numerical simulations and experiments, opens the possibility for alternative approaches to the design of compliant actuators, whereby rendering of high stiffness is possible if the load mass is always higher than a determined threshold.

Keywords: Compliant actuators, haptics, coupled stability, force control

1. Background

The possibility of safe physical interaction and successful cooperation with humans is among the most promising and exciting frontiers of robotics. Several
new scenarios such as rehabilitation robotics [1, 2, 3], human augmentation robotics [4], surgical robotics [5, 6] and haptics have rapidly transitioned from
science-fiction, to research laboratories to flourishing industries. In all those scenarios, a common underlying feature is the need of regulating the physical

⁸ interaction between a human and a robot.

In order to achieve this goal, interaction control approaches, as described 10 in the impedance control framework [7], have extensively been implemented, and their effectiveness demonstrated in several applications requiring physical

- ¹² interaction with humans. In the case of simple impedance control [7], [8], the controller is defined in impedance causality, and the mechanism is modeled as
- ¹⁴ a pure source of effort variables. Successful implementation of this type of impedance controller has been demonstrated mostly in the case of robots with
- negligible intrinsic dynamical properties or whose motion is approximated by quasi-static movements [1, 9, 3].

The problem of accurately regulating interaction becomes more difficult in the case of manipulators with complex dynamics, with high inertia, and/or with

- highly non-transparent actuation systems [10]. When robots are intended for applications requiring substantial assistance to humans during load-intensive
- tasks, manipulators are indeed not pure effort sources, due to the lack of actuation systems that allow achievement of high force density without substantial
- ²⁴ increase of task-space dynamic loading. In this case, model-based dynamic compensation schemes for impedance control can be adopted [11], but often
- 26 do not fully guarantee accurate interaction control. In general, effectiveness of model-based schemes is limited by neglect of higher order or nonlinear dynami-
- cal effects, and by issues related to sensorization and practical limitations in the capabilities of full state feedback, required for compensation of inertial loads.
- ³⁰ Force feedback, an approach pursued since the late 1970s [12], can enable the accurate regulation of interaction also in non-transparent manipulators. In
- ³² such architectures, an explicit measurement of the force of interaction between the manipulator and the environment is used to generate a command signal,
- that ultimately regulates interaction with the environment. Since then, several forms have been proposed for the force controller, such as proportional control,
- ³⁶ pure integral control, proportional-integral (PI) control [11], and controllers with inner motion-control loops ([13, 14] for compliant joints, and [15] for rigid
 ³⁸ actuators).

1.1. Passivity for force-feedback systems

- ⁴⁰ Real-world implementations of interaction control through force-feedback do not succeed in achieving arbitrary impedance values. In practice, when a con-
- ⁴² troller attempts to emulate dynamics that differ significantly from those of the hardware, the risk of instability increases [16]. The stability limits in force-
- feedback controlled systems have been approached through the concept of passivity, a concept adapted from classical theory of electrical networks [16]. It has
- ⁴⁶ been proved that when two stable systems with passive impedance port function are coupled together, the coupled system, that results from the connection of

- ⁴⁸ the two systems, is stable. Instead, if a robot is stable but non-passive, there will be at least a passive environment that, during interaction, will destabilize
- the controlled system [16, 17]. Proving passivity of a controlled system ensures stability for a wide range of interaction environments that include human dy-
- ⁵² namics, that are generally modeled as a passive, non linear, first- or secondorder system. Also, passivity can be proved for a simplified model, that in-
- ⁵⁴ cludes only the controlled system, and do not require detailed knowledge of the environment. Despite the introduced simplicity, this approach allows derivation
- of strong conclusions on the stability properties of the robot, when coupled to an extremely large and useful set of environments.
- Requiring passivity has also drawbacks. Colgate showed that if endpoint force feedback is used to compensate for also the distal mass, the system becomes
- ⁶⁰ non-passive [17]. This limitation results from the often unavoidable presence of dynamics between the force sensor and the actuator, that can severely limit the
- ⁶² performance of force and impedance controllers. Achievement of global passivity properties for a controller (i.e. passivity for all frequencies) is considered
- an important requirement for human-interacting robots; yet, at the same time, it is also acknowledged to be quite conservative [18]. It is indeed recognized
- that the frequency-domain passivity requirement often poses excessive limitations on performance. Examples of approaches violating the frequency-domain
- passivity requirement without limitations of coupled stability are time-domain passivity controllers [18], also implemented in teleoperation systems [19], and
- ⁷⁰ force-feedback controllers for wrist robots[20].

1.2. Inclusion of physical compliance for interaction control

- ⁷² An alternative approach to improving the reflected dynamics of manipulators requires considerable changes to actuators design, as done in Series Elastic
- Actuators (SEA)[21, 22], where a compliant element is introduced in series between the actuator and the load and its deflection measured. This measurement
- r6 enables estimation of the interaction forces exchanged between the actuator and the environment, and ultimately of the interaction forces between the human

⁷⁸ and the robot.

SEAs were originally proposed for their mechanical advantages over stiff actuators, such as shock tolerance and increased power capabilities [21, 23, 24]. 80 In later years, several research groups showed that Series Elastic Actuators can be successfully employed for accurate implementation of interaction control 82 approaches with actuation systems that could not be modeled as low-impedance effort sources [25, 26, 27]. From a control perspective, the primary advantage of 84 series elasticity over stiff force feedback is that the compliant force sensor reduces the physical gain of the feedforward path in the force control loop. In this way, 86 the control gain can be proportionally increased to maintain the overall loop gain of the actuator, resulting in the same stability margins with higher control 88 gains [22]. SEAs can then display a lower output impedance than the one of the actuator alone, without imposing the same stringent limits on the maximum 90 reduction of endpoint inertia as is the case for systems with stiff force-feedback. In fact, this scheme allows simultaneous adoption of high-geared motors and 92

achievement of low apparent load inertia, since motor inertia is decoupled from ⁹⁴ the load through the series elastic element.

Though the requirement on minimum reflected inertia for coupled stability does not apply to SEAs, it has been demonstrated that, if inner velocity loops are introduced in the control architecture, the SEA is not passive if it attempts to

⁹⁸ regulate a behavior corresponding to a pure spring with elastic constant higher than that of the physical compliant element [14]. In the following, we will refer

to this case as "virtual stiffness control", in which it is desired to regulate the force of interaction F_i , in a way that it is proportional to the error between the

¹⁰² measured position x and a desired position x_{des} through a constant K_{des} , named virtual stiffness, so that $F_i = K_{des}(x_{des} - x)$. Although different passivity con-

ditions are obtained through alternative controllers types, the presence of inner velocity loops is often mandatory. This is certainly the case of non-backdrivable

¹⁰⁶ actuators which can be more suitably modeled in admittance causality, such as piezoelectric actuators. For this class of actuators, cascaded force-velocity

¹⁰⁸ represents the most direct implementation of an interaction controller. Unfor-

tunately, the limit on passivity (achieved only for $K_{des} \leq k_s$) poses a stringent limitation on the maximum stiffness that can be accurately rendered through a compliant actuator that includes a non-backdrivable motor. Practical exam-

- ples of such velocity-sourced motors are ultrasonic piezoelectric motors, such as the one commercialized by Shinsei and utilized in MR-compatible robotics
- ¹¹⁴ applications [28, 29].

This paper investigates the consequences arising from the violation of the passivity requirement, when the actuated system is controlled to render a pure virtual stiffness with elastic constant higher than the physical spring of the

¹¹⁸ SEA. The analysis is conducted for non-backdrivable motors that are modeled as ideal velocity sources, for which it is possible to derive the parameters of a

stiffness compensator capable of achieving coupled stability with a wide range of passive environments. The approach is finally validated through both numerical
 simulations and experiments in a 1-DOF test bench.

2. Modeling and problem definition

- The schematic of a SEA is presented in Fig. 1, in which an actuator drives the output mass through a spring-mass-damper system. The compliant actuator regulates the force of interaction with the environment F_L through measurement of x_M and x_L , representing motor and load displacements, respectively. Assuming hyperbolic draw of the gravity placetic constant h_L F_L is estimated through the placet
- ing knowledge of the spring elastic constant k_s , F_L is estimated through Hook law: $F_L = k_s(x_L - x_M)$.



Figure 1: Mechanical schematic of a Series Elastic Actuator. A motor is connected to the load through a spring, whose deflection is measured, thus allowing measurement of the interaction force F_L .



Figure 2: Block diagram of the controlled system. A non-backdrivable motor is controlled to be a velocity source, and an outer loop is closed on the measured force of interaction with the environment. A velocity control disturbance transfer function can be defined to describe the effect of interaction force on error in velocity control. This function will be neglected, assuming negligible admittance for the velocity controlled motor.

The goal is to develop a controller which is able to regulate the apparent 130 impedance $Z_L = \frac{F_L}{\dot{x_L}} = \frac{k_s(x_L - x_M)}{\dot{x}_L}$ of the system at the port of interaction with the environment, (i.e. at coordinate x_L), through only measurement of the 132 two positions x_L and x_M . In particular, we are interested in the possibility of rendering the behavior of a pure spring, with virtual constant K_{des} higher 134 than that of the physical spring, k_s . The environment is defined by the mass m_L in the figure, but it could be any linear passive environment, (i.e. any 136 combination of spring and mass). To this aim, we will investigate whether there are control actions that guarantee passivity of the transfer function Z_L , that 138 would consequently imply stability of the system when coupled to any passive environment. 140

3. Cascaded force-velocity control

¹⁴² Several control approaches have been proposed for force control of SEAs, including direct feedback force controllers with feedforward compensation [30],

- ¹⁴⁴ nonlinear compensators capable of reducing the effects of friction and variability of interaction dynamics [31] and cascaded linear force-position [32] or force-
- ¹⁴⁶ velocity control [33, 13, 34]. In the specific case of SEAs designed to improve the interaction control performance of non-backdrivable or high impedance ac-

tuators, the latter two controllers are of particular interest, since they effectively allow the conversion of a force control problem into simpler position or velocity

¹⁵⁰ control problems, thereby enabling the implementation of interaction controllers for non-transparent motors. The cascaded force and velocity control scheme was

demonstrated to be passive [34] for a wide range of desired impedance values. This holds true even in the absence of viscous friction, which is not generally

¹⁵⁴ the case of direct force/torque feedback control [35].

Fig. 2 reports a general block diagram of the cascaded force-velocity controller for an SEA in the Laplace domain, C_Z and C_F being the transfer functions of the stiffness and force controller, respectively. The inner velocity loop

can be modeled by the superposition of two contributions, one describing velocity control performance, and the other describing the degradation of velocity

¹⁶⁰ control due to the interaction with the environment:

$$V_M(s) = H_v(s)V_{des}(s) + D_v(s)F_L(s),$$
(1)

where H_v is the velocity control closed loop transfer function, in the absence of torque disturbance, and D_v describes the effect of torque disturbance on velocity control output.

For non-backdrivable motors, within their linear range, the effect of velocity reduction provided by interaction with the environment is negligible. Such simplification is accurate for a class of actuators such as piezoelectric actuators, which have a very small intrinsic admittance both in their unpowered and in their velocity-controlled modes [36]. In order to model this effect, we will assume perfect disturbance rejection from the velocity controller, and we will assume that the term $D_v(s)F_L(s)$ is negligible compared to the first term in (1), implying no effect deriving from load torque on the motor velocity control

¹⁷² performance.

Under the mentioned assumptions, the velocity-controlled ultrasonic motor has been modeled as an ideal velocity source, as described in the block diagram shown in Fig. 2, assuming that $D_v = 0$. In order to regulate interaction, an outer loop is then closed on the measurement of the load position x_L , which is compared to the desired load position, in order to define a desired force value

 F_{des} , determined as a stiffness force field converging towards the desired position x_{des} . In interaction control, it is often desired to regulate mechanical interaction

between subjects and the robot by specifying different values of virtual stiffness K_{des} , thereby regulating the level of mechanical assistance towards the equilib-

rium position provided by the robot. In the following, we will analyze which are the limits of virtual stiffness values that can be rendered in a stable manner,
and design control actions to improve stability and accuracy of stiffness control.

3.1. Limitations of pure stiffness control through cascaded force-velocity control

The most straightforward way to implement a stiffness controller through the block diagram shown in Fig. 2 is to impose $C_Z = 1$. Correspondingly, the outer stiffness loop will be commanding elastic desired interaction forces, displaying a force field proportional to the difference between the desired load position and the measured load position, with an arbitrary stiffness constant K_{des} . In this case, using a proportional controller as force compensator (i.e. $C_F = k_{p,f}$), the impedance of the controlled system Z_L can be calculated as:

$$Z_L = \frac{k_s s + (K_{des} k_{p,f} k_s)}{s^2 + (k_{p,f} k_s)s}$$
(2)

Analysis of the passivity of the controlled system impedance transfer function allows determining the coupled stability during interaction with passive environments. In order for the controlled system to be passive, the impedance transfer function Z_L needs to be stable and to satisfy the condition $Re(Z_L(\omega)) > 0, \forall \omega \in$ R. Through symbolic calculation, assuming $s = j\omega$, we can calculate the real part of the Z_L transfer function, and evaluate conditions for it to be positive, hence concluding on the passivity of the system. We obtain

$$\operatorname{Re}(Z_L(j\omega)) = \frac{k_s k_{p,f}(k_s - K_{des})}{\omega^2 + (k_{p_f}^2 k_s^2)},$$
(3)

which is clearly positive only for $k_s > K_{des}$, for all values of the force compensator gain. The same result can be derived for a force compensator that includes a proportional-integral controller, as the one proposed for passive interaction control of SEAs [14]. Hence, the system is not passive if it is commanded to ¹⁹⁰ render a virtual spring with stiffness higher than that of the physical spring.

4. Coupled stability and performance during display of high-stiffness environments

The analysis of the Bode plot of the impedance transfer function of the 194 controlled system is shown in Fig. 3. It can be seen that when $K_{des} > k_s$, the 195 impedance transfer function approximates a pure spring at low frequencies, and 196 becomes a non-passive transfer function (with phase lower than -90 deg) above a certain frequency value, that can be modulated through action on the force

¹⁹⁸ feedback compensator gain.

192



Figure 3: Bode diagram of the apparent impedance of the system, when controlled through the proportional impedance compensator, $F_{des} = -K_{des}(x_L - x_{des})$, with a proportional force feedback compensator $C_F = k_{p,f}$.

In order to guarantee stability during rendering of a high-impedance environment, we relax the constraints on the specific form of force and stiffness compensators, and conduct an analysis in which we will assume them to be generic, causal compensators C_F (for the force compensator), and $K_{des}C_Z$ (for the stiffness compensator), and consider a causal rational polynomial expression for the velocity control transfer function H_v . The virtual impedance Z_L rendered at the load side, is given by:

$$Z_L(s) = \frac{k_s}{s} \frac{s + K_{des} C_F H_v C_Z}{s + k_s C_F H_v} = \frac{k_s}{s} K(s)$$

$$\tag{4}$$

Analysis of (4) provides significant insight on the behavior of the system. In particular, the first term defines the physical properties of the system reflected to the output: a pure spring with stiffness k_s . In contrast, the second term, labeled as K(s), is a transfer function that describes how the physical stiffness is modulated to be displayed to the output. If K(s) = 1, the system displays a stiffness exactly equal to that of the physical spring. Modulations in the amplitude of K(s) allow regulation of different stiffness values. If the phase of K(s) is positive, this will add damping to the system, whereas if the phase of K(s) is negative, this will contribute to further decreasing the phase of the impedance transfer function Z_L below its value of -90 deg dictated by the pure

- ²¹⁶ integral action of the spring. In particular, it should be noted that, unlike the case of servo controls, the transfer function of force and impedance compensators
- do not enter linearly into the open-loop transfer function $Z_L Y_e$ (that determines stability of the system when coupled to the environment with admittance Y_e),
- ²²⁰ nor into the transfer function Z_L (that determines performance of the haptic display). This limitation is, in general, inherent to interaction controllers [10]
- that operate without a specific model of the environment. This makes design of a controller aimed at regulating interaction with an unknown environment a
- ²²⁴ difficult task that cannot be addressed with traditional methods borrowed from the servo controllers literature.
- 226

It is useful to analyze how the conditions for accuracy and passivity translate into requirements for the K(s) transfer function. For pure stiffness control, it is

desired to render a pure spring, with stiffness K_{des} . In this case, the performance condition is:

$$Z_L(\omega) = \frac{K_{des}}{j\omega}, \omega \in \mathbf{\Omega}_c, \tag{5}$$

where Ω_c is a range of frequencies of interest, for which it is desired to accurately regulate interaction. The condition for passivity of the haptic display dictates that the apparent stiffness fraction K(s) needs to satisfy the following relation, $\forall \omega$:

$$0 < \arg K(\omega) < \pi \tag{6}$$

We now evaluate how the choice of compensators C_F and C_Z allows ful-230 fillment of coupled stability during interaction with the environment. We will focus on the case $K_{des} > k_s$, which is a challenging condition for which to ensure 232 coupled stability. In fact, it can be intuitively demonstrated that no choice of causal, finite-magnitude compensators C_F and C_Z allows fulfillment of the pas-234 sivity requirement with the cascaded force-velocity control scheme. In fact, if C_F and C_Z have finite magnitude response at all frequencies, $\lim_{s\to\infty} K(s) = 1$, which 236 implies that at high frequencies, the apparent stiffness of the system reduces to the physical stiffness of the series elastic element. However, for the case in 238 which the desired stiffness is higher than the physical stiffness of the spring (i.e. $K_{des} > k_s$), the modulus of the stiffness transfer function |K|, within the range 240 of frequencies Ω_c will be such that $|K(\omega_1)| > |K(\omega)|, \forall \omega_1 \in \Omega_c, \omega > \max(\Omega_c)$. Hence, for $K_{des} > k_s$, the imposition of the performance requirement implies 242 that K(s) is required to have a flat, zero-phase region at low frequencies, with amplitude $K_{des}/k_s > 1$, and a second flat, zero-phase region (above the fre-244 quency band Ω_c where performance is specified), of unitary amplitude. Due to a fundamental property of Bode plots, the two zero-phase regions (higher 246 amplitude at lower frequencies and lower amplitude at higher frequencies) can not be simultaneously present in a system, without the phase diagram having 248 at least one point with negative phase. It is thus not possible, with the reported control approach, to render accurate, zero damping, haptic display, with 250 stiffness $K_{des} > k_s$, respecting the passivity requirement. The mentioned limitation has profound consequences: from the analysis of the closed-loop transfer 252 function $Z_L Y_e$, it can be demonstrated that, subject to proportional force conthe system is unstable when coupled to any environment consisting of a mass $m_e > 0$ (i.e. the worst possible coupled stability result).

4.1. Role of force compensator on coupled stability

Not every choice of force and stiffness compensators C_F and C_Z implies coupled instability with every value of environment mass. We will demonstrate this through two separate analyses that consider different choices of compensators C_F and C_Z . Let us first assume the following structure for the force compensator:

$$C_F = k_{p,f} \frac{s + z_F}{s + p_F} \tag{7}$$

Through this expression, the force compensator is expressed by a transfer function with one zero and one pole that can take the form of a lead compensator (for $z_F < p_F$), a lag compensator (for $z_F > p_F$), a proportional controller $p_F = z_F$, a proportional-integral action (for $p_F \to 0$) and a causal derivative action (for $z_F \to 0$). For this controller, the impedance transfer function Z_L^f obtained under the effect of the force compensator is:

$$Z_L^f(s) = \frac{k_s s^2 + (k_s p_f + K_{des} k_{p,f} k_s) s + K_{des} k_{p,f} k_s z_f}{s^3 + (p_f + k_{p,f} k_s) s^2 + (k_{p,f} k_s z_f) s},$$
(8)

which, in agreement with the general demonstration presented in section 4, is 258 not passive for any choice of the control parameters $k_{p,f}$, p_f and z_f , when $K_{des} > k_s$. It is interesting to observe the coupled instability of the system, 260 when it is interacting with its most destabilizing environment. It can be verified that for this stiffness controlled system, the most destabilizing environment 262 is a pure mass, i.e. the admittance $Y_e = \frac{1}{m_e s}$. In this condition, the closed-loop system is unstable for every value of interacting load mass $m_e > 0$, for every 264 value of the controller parameters. In fact, through calculation of the characteristic polynomial $CLP = 1 + Z_L^f Y_e$ and application of the Routh-Hurwitz 266 stability criterion to the coefficients of its numerator, it is possible to simplify the resulting system of inequalities to the simple expression: $K_{des} \leq k_s$. 268

This demonstrates that coupled stability can be obtained only for values of desired stiffness less than or equal to the stiffness of the physical spring, regardless of the specific choice of the force feedback compensator (i.e. a lead

²⁷² or lag compensator, PI controller or compensator with a derivative action), for every value of interacting mass.

274 4.2. Coupled stability through stiffness compensator

The simplest method to guarantee stability of the system is through an explicit damping action by the controller, that can be expressed in the form of a velocity-dependent action of the stiffness compensator C_Z . This can be obtained, for example, by employing proportional force feedback, and considering the following form for the stiffness controller: $C_Z = 1 + \frac{B_{des}}{K_{des}}s$. In this case, the resulting system can be made passive for certain values of desired stiffness K_{des} . In fact, the impedance transfer function Z_L^b , under the action of the damping action is:

$$Z_L^b = \frac{(k_s + B_{des}k_{p,f}k_s)s + K_{des}k_{p,f}k_s}{s^2 + k_{p,f}k_s s},$$
(9)

which is passive also when $K_{des} > k_s$, if the controller is introducing an amount of damping higher than the threshold B^* , defined as

$$B^{\star} = \frac{K_{des} - k_s}{k_{p,f} k_s}.\tag{10}$$

Some practical limitations arise from the adoption of this controller. Fist, if no direct measurement of velocity is assumed, the anti-causal damping action cannot be practically implemented in real systems. Practically, the damping term will be implemented as a causal derivative action, which behaves as a pure differentiator¹ at low frequencies, and reduces to a proportional gain at higher frequencies, as provided by the causal differentiator transfer function $G = \frac{Ns}{s+N}$. Introducing the non-ideal differentiator transfer function in the impedance compensator, we then lose again the possibility of proving passivity for the impedance-controlled system for every $K_{des} > k_s$. Indeed, the effect

¹We are neglecting the effect of the required low-pass filtering and delay that is needed to differentiate digital position signals deriving from encoders, that are affected by quantization.

of a non-ideal compensation term is significant. Because of the introduction of the damping B_{des} , it can be shown that for every finite value of N, coupled

instability will occur for every value of environment mass $m_e > 0$, i.e. the worstpossible coupled stability result. This issue can be mitigated by introducing damping in the system in excess, compared to what is required in theory in

296 damping in the system in excess, compared to what is required in theory in (10). However, this solution would reduce the accuracy of haptic display also

at low frequencies, where in theory no damping action would really be needed for coupled stability. In order to address the limitations of this pure damping
action, we will then investigate the effect of the introduction of lead and lag in the stiffness compensator, by defining it in the form²:

$$C_Z = \frac{p}{z} \frac{s+z}{s+p} \tag{11}$$

In this case, the resulting impedance transfer function, under the effect of the stiffness compensator, Z_L^k is:

$$Z_L^k(s) = \frac{k_s z s^2 + (k_s p z + K_{des} k_{p,f} k_s p) s + K_{des} k_{p,f} k_s p z}{z s^3 + (p z + k_{p,f} k_s z) s^2 + (k_{p,f} k_s p z) s}$$
(12)

Again, the introduction of an amplitude-bounded causal compensator C_Z , acting on the error between measured and desired load position, results in an impedance transfer function that is passive only if $K_{des} \leq k_s$. However, an analysis of coupled stability of this controller provides results that allow for margins in controller design. In fact, through evaluation of the closed loop polynomial during interaction with the most destabilizing environment (again, a pure mass), and application of the Routh-Hurwitz stability criterion, it can be seen that coupled stability can be obtained for a wide range of interacting mass values, that can be modified by a proper choice of controller gains. Reduction of the Routh-Hurwitz determinants provides the following conditions for coupled

312 stability $(m_e > 0)$:

²The symbolic form of the compensator C_Z is similar to the one chosen for C_F , but an amplitude normalization factor $\frac{p}{z}$ has been introduced, so as to guarantee that the stiffness displayed at low frequencies matches the desired stiffness K_{des} .

1)
$$m_e < m^* \text{AND } p < \frac{k_s z}{K_{des}} \text{AND } \left[k_{p,f} \neq \frac{K_{des} z - k_s z}{K_{des} k_s} \right]$$

2)
$$m_e > m^* \text{AND } p > \frac{K_{des} k_{p,f} k_s z}{K_{des} k_{p,f} k_s - K_{des} z + k_s z}$$
(13)

with
$$m^* = \frac{k_s (K_{des} k_{p,f} + z) (K_{des} p - k_s z)}{z (k_{p,f} k_s + p) [p z k_s + K_{des} (k_{p,f} k_s (p - z) - p z)]}$$

showing that coupled stability can be obtained as a function of a threshold load mass value m^* . Though this controller cannot be proved stable with arbitrary values of environment mass m_e , it allows modulation of the value of m^* to ensure that the value of admissible environment mass m_e is always within the stability range defined by (13). In particular, the value of m^* can be reduced to guarantee that the second inequality shown in (13) is respected in all operating conditions. Through this control design method, it is possible to guarantee that the system is stable for values of interacting mass higher than a certain threshold.

This approach can be surprisingly useful, since in most SEA designs, the 322 minimum value of interacting mass m_e is never equal to zero due to inherent design constraints (e.g. a linear SEA will have a load plate that will be used 324 to connect to a load; a rotary SEA will be necessarily connected to some output link, whose inertia adds to the load inertia). Moreover, for applications 326 that involve wearable human-interacting robots, such as rehabilitation or human augmentation, the minimum value of mass connected to the load is often 328 significant. For example, the mass might consist of the inertia of the hand for a wrist exoskeleton, or of the inertia of the lower leg for a knee exoskeleton. 330 Using a constant gain controller, the system can be made stable with masses whose value can be seen in Fig. 4, parameterized in terms of force feedback 332 gain and desired stiffness value. It can be seen that this control approach al-

- ³³⁴ lows achievement of coupled stability with a limited set of environments, that can be regulated through control parameters, in a way that the range of ad-
- missible values for coupled stability is well contained in the range of interacting environments that can be considered in a given application.

We now investigate which values of minimum and maximum interacting mass m^* that guarantee coupled stability can be obtained with feasible combinations

³⁴⁰ of control parameters. In particular, we conduct this analysis by assuming a fixed value for the force compensator control gain, and investigate the de-

pendency of m^* on the characteristics of the impedance compensator gain, for different values of desired stiffness.

The results of this analysis are shown in Fig. 4, where the regions of compensator values resulting in minimum and maximum values of interacting mass

that guarantee coupled stability are contoured by dashed lines. In particular, it can be seen that pure stiffness control p = z would result in coupled instability

³⁴⁸ for every value of interacting mass (dark red region interposed between the two colormap regions where stability is possible for some values of mass).

From the coupled stability standpoint, it would seem that the best choice of C_Z could either be that of a lead-compensator, with one pole and one zero in the high frequency range, or that of a lag compensator, with low-frequency pole and zero. In fact, in the first case (z < p), coupled stability is achieved for values

of environment mass higher than a threshold, that can be decreased through control gains (i.e. moving p towards higher frequencies and increasing the force-

feedback compensator gain $k_{p,f}$). In the second case (p < z), coupled stability is achieved for values of environment mass lower than a threshold, which can be

again increased through control design (i.e. moving z towards lower frequencies and increasing the force-feedback compensator gain).

Stability is not the only requirement in the design of interaction controllers. Although the stiffness compensator C_Z can be designed to have unitary magni-

- tude and zero phase at low frequencies (resulting in an accurate, pure-stiffness interaction behavior at those frequencies), controller parameters can influence
- the upper limit of this frequency region. To this aim, we define a measure of interaction controller accuracy, as described in [10], and include this measure-
- ³⁶⁶ ment within an optimization framework. The framework would optimize the combination of control parameters so as to maximize both performance and
- stability, that for this specific controller could be defined by the range of masses



Figure 4: Logarithmic color maps showing values of threshold environment mass m^* , for coupled stability during stiffness control at different values of K_{des} (columns), for different values of force compensator gains, $k_{p,f} = 1$ (a), and $k_{p,f} = 10$ (b). Use of lag compensators (p < z - upper-left triangular section of the p - z plane) result in coupled stability for values of interacting mass $m_e < m^*$, while lead compensators (p > z - lower-right triangular section of the p - z plane), result in coupled stability for values of interacting mass $m_e > m^*$. The region where coupled stability is not guaranteed for any value of interacting mass is separated by the dashed line-contoured regions and is reported in dark red.

that result in a stable interaction.

Under the "ideal non-backdriveability, ideal velocity control" hypothesis stated so far, the system has been simplified to the point that it lends itself to an intuitive, straightforward and informative performance analysis. In fact, from the analysis of the K(s) transfer function, it can be seen that the effect of the impedance compensator is to introduce a phase lead or lag starting at the frequency where the lowest frequency pole or zero of the impedance compensator is located. This can be seen also from the analysis of the impedance transfer function, parameterized as a function of the specific value of compensator gains



Figure 5: Resulting impedance transfer function Z_L^k , when p = 1 and for different values of z, for a unitary force feedback compensator gain and a desired stiffness $K_{des} = 2k_s$. Circled points are those in which the magnitude of the modulated stiffness transfer function K(s) is attenuated by 3 dB. The frequencies in which this situation occurs are in correspondence with the minimum pole or zero of the impedance compensator.

(that change from a lead compensator to a lag compensator). The analysis can be conducted by keeping the pole location constant, and observing the effect on the resulting impedance transfer function, as a function of the zero location, as reported in Fig. 5. It can be seen that accurate impedance rendering is obtained for $\omega \in (-\infty, \min(p, z)]$, i.e. up to a frequency that depends on the minimum value zero or pole introduced in the compensator. A very simple performance measure can be introduced for this system, as given by the two-dimensional function acc, defined as:

$$\operatorname{acc} = \min(p, z),$$
 (14)

³⁷⁰ with its colormap reported in Fig. 6.

From an intersection analysis between the two plots, it is now clear that an acceptable compromise between stability and performance is provided only by a control action of a lead-compensator type (i.e. p > z). In fact, this controller allows achievement of accurate interaction up to a frequency $\omega_c = z$, and guarantees coupled stability for interaction masses higher than a certain threshold, that can be reduced through control action to values that are largely



Figure 6: Measure of accuracy for the interaction controller implemented through the impedance compensator C_Z , determined as $\max(p, z)$, in a logarithmic scale. The plotted function represents a quarter of a square pyramid (the lower-left quarter of the pyramid, from the top view shown in the colormap), cut through two orthogonal planes that contain the pyramid axis and one of the axes defined by the basis edge, respectively. The pyramid vertex, i.e. the point $p = p_{MAX}$, z = p, corresponds to the point with highest accuracy.

lower than the realistic values associated with a given application.

In these conditions, it can be verified by inspection of Eq. (13) that the value of minimum mass required for a stable interaction is linear with the value of virtual stiffness. This means that, once control gains have been tuned, the higher the value of desired stiffness K_{des} , the higher the minimum load mass in

4.3. Effect of a non ideal inner velocity loop

We now address the effect of model non-idealities, introducing a high-frequency attenuation in velocity control performance, modeling the velocity control transfer function H_v as a low-pass filter $\frac{1}{s\tau+1}$. We describe the effect of τ on both the accuracy and performance of interaction control. First, we analyze the effect of the inner velocity loop transfer function on the selection of the optimal controller gains that maximize stability and evaluate the effect on the haptic display accuracy. For every set of controller gain values, the minimum value of interacting mass for coupled stability is computed through a root locus analysis, analyzing

³⁸² order to achieve coupled stability.

the interaction with the most destabilizing environment (a pure mass). This can be done by calculating the minimum value of m_e that results in the admittance environment transfer function $Y_e = \frac{1}{m_e s}$ such that the polynomial *CLP*, that describes the stability of the coupled system, with

$$CLP = 1 + Z_L Y_e \tag{15}$$

has roots in left half complex plane. Also, for every set of controller gain values, a measure of impedance control accuracy is introduced as in [10], as the reciprocal of the cost function, calculated in a logarithmically-spaced set of frequencies of interest $\Omega = \{\omega_1, \omega_2, ..., \omega_n\}, \omega_1 = 10^{-3}, \omega_n = 10^3$:

$$C(\boldsymbol{\omega}) = \sum_{i=1}^{n} |\log |Z_L(j\omega_i)| - \log |Z_{des}(j\omega_i)||, \qquad (16)$$

that compares the amplitude of the resulting impedance transfer function Z_L to the desired impedance Z_{des} . The analysis is reported in Fig. 7 for specific values of K_{des} and τ , varying controller gains. The analysis shows that the region of controller gains that maximizes accuracy is, in agreement with the results obtained in the ideal velocity control case, the upper-right corner of the p-z plane, i.e. the region that maximizes p and z, with p > z. The introduction of the non-ideal velocity controller slightly modifies the optimality conditions for coupled stability. In fact, minimization of the environment mass for coupled stability is obtained when the compensator zeros/poles are in a lower frequency

range. The specific value of the upper limit of this frequency range will dependon the value of the velocity loop transfer function time constant.

To further investigate the issue of optimal zero-pole location in the impedance ³⁹⁶ compensator C_z , we addressed both stability and accuracy, and evaluated how such properties vary as a function of inner velocity loop time constant τ . Once

- the controller parameters p_{opt}^{τ} and z_{opt}^{τ} that yield the minimum value of interacting mass for coupled stability are determined for every value of τ , the accuracy
- measure 1/C^τ_{opt} of the impedance-controlled system is calculated, and reported (see Fig. 8; labeled as Opt_τ). Similar analysis was conducted for the ideal
 compensator gains obtained for the ideal velocity loop case (labeled as Opt),



Figure 7: Effect of stiffness compensator parameters on the minimum value of interacting mass for coupled stability ($m_{e,MIN}$, shown in the top plots) and on the accuracy of the impedance transfer function, defined as 1/C - lower plots, for frequencies logarithmically spaced between 10^{-3} rad/s and 10^3 rad/s. An inner velocity loop modeled as a first-order low-pass filter with time constant $\tau = 0.01$ is considered in this analysis, and the value of the force-feedback compensator gain is set to 1 (left) and 10 (right).

and for the cases in which the minimum damping B^* , as expressed in eq. (10), ⁴⁰⁴ is introduced in the compensator (labeled as B^*), or increased damping equal to $2B^*$ was introduced, labeled as $2B^*$. The analysis is graphically reported ⁴⁰⁶ in Fig. 8, and enables description of the achievable trade-off between stabil-

⁴⁰⁸ non-backdrivable actuators, controlled as velocity sources, as a function of the responsiveness of the inner velocity loop.

ity and accuracy of high-stiffness impedance control in compliant robots with

⁴¹⁰ First, it can be seen that the accuracy and coupled stability requirements are



Figure 8: Effect of different inner velocity loop time constant τ on the accuracy and stability of the impedance controlled compliant actuator. Circle-marked dashed lines represent the performance measure in the left logarithmic scale, while diamond-marked continuous lines represent the stability measure, in the right logarithmic scale.

both negatively influenced by slower inner velocity loops. For all considered controllers, a faster inner velocity loop is able to simultaneously provide increased accuracy within the considered frequency range, and lower values of minimum
interacting mass for coupled stability (increased stability performance).

A full quantitative comparison between the designed compensator and the system controlled through a pure damping action where the minimum damping B^* is injected in the system is not possible, due to the fact that such a controller

results in an unstable system, for all values of interacting mass $m_e > 0$ (due to consideration of frequency-bounded controller derivative action). For this

 $_{\rm 420}$ $\,$ reason, it is not possible to define a measure of stability of this controller.

From a comparison between the results achievable with the two different 422 controllers (i.e. Opt vs. Opt_{τ}), it can be seen that the solution found through separate optimization for different values of τ (Opt_{τ}) provides the best coupled

stability results for every value of τ , compared to the controller designed for the 424 ideal velocity loop case (Opt). The amount of reduction in values of minimum

mass that provides coupled stability (Δm_e) allowed by the Opt_{τ} controller is 426 very small at low values of τ , $(\frac{\Delta m_e}{m_e} \leq 5\%, \tau \leq 0.01)$, and becomes more significant while τ increases, settling to $\Delta m_e = m_e$, for the case in which $\tau = 1$. 428

It can be seen that at low values of velocity control time constant ($\tau < 0.01$), coupled stability can be achieved with very low values of interacting mass, at 430 the expense of haptic display accuracy (the minimum value of interaction mass is reduced in the Opt and Opt_{τ} controllers to less than half of what is allowed 432 with the $2B^{\star}$ controller, but accuracy is reduced by a similar amount). This results from the fact that the range in which control parameters p and z are 434 sought is $[10^{-3} \ 10^3]$ rad/s, representing realistic hardware limitations in realtime control applications. In general, it is shown that the performance obtained 436 through the controller determined for the ideal velocity loop case (Opt) looses its optimality when the region in the frequency domain in which poles and zeros 438 of the outer loop compensator are sought $([10^{-3} \ 10^3] \ rad/s)$ overlaps with the region at which the velocity control performance degrades.

5. Model validation

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The feasibility of the developed controller is validated through numerical 442 non-linear modeling and simulations, as well as through experiments in a compliant actuator (MR-SEA II) that includes a non-backdrivable traveling-wave 444

ultrasonic motor, a cable transmission and two pre-loaded extension springs.

5.1. System description and modeling 446

The 1 DOF actuator prototype (MR-SEA II) was developed for an MRcompatible haptics application [36] and comprises a rotary ultrasonic piezoce-448 ramic motor (5W Shinsei ESR60-E3N), a threaded pulley on the motor shaft, a

cable transmission, pre-extended phosphor bronze extension springs, a custom-450 designed Delrin carriage and a linear ceramic balls bearing, supporting the



Figure 9: MR-SEA II. (1) phosphor bronze extension spring, (2) ceramic linear ball bearings, (3) piezoceramic motor, (4) rotary optical encoder, used to measure x_M , (5) linear encoder, used to measure x_L , (6) cable transmission, (7) slider, with plastic eyebolts.

carriage. As shown in the companion design paper [36], the motor is non-452 backdrivable and when controlled through its own factory-tuned velocity controller can be modeled as a low admittance velocity source, and compensated to 454 regulate velocity linearly (i.e. without amplitude dependence, except for its low velocity region) within its operating frequency and amplitude range (see Table 456 1). A numerical model is implemented in Simulink (the Mathworks Inc.), and run with an ODE 45 solver at a fixed step of 1 ms, to validate the theoretical 458 analysis assessing the effects of model non-linearities on the closed-loop stability of the system. In the model, the actuator is controlled to display a stiffness 460 K_{des} through the action of the impedance compensator C_Z acting on the cascaded force-velocity controller, with desired position $x_L = 0$. This controller 462 action responds to a load force perturbation, modeled as a step with amplitude of 4 N (20% of the actuator force capabilities). The case of force perturbation 464 when the compliant actuator is interacting with an environmental mass m_e is chosen because it closely matches the task conducted during the experimental 466 characterization, and because we have demonstrated that it corresponds to the

Table 1: MR-SEA performance

Property	Value	
ine Maximum continuous force	20 N	
Spring stiffness	3.8 N/mm	
Maximum velocity	$0.1 \mathrm{~m/s}$	
Velocity control frequency range	0 - 10 Hz	
Motor encoder quantization	$1.3 \cdot 10^{-2} \text{ mm}$	
Load encoder quantization	$1 \cdot 10^{-2} \text{ mm}$	
Load mass	100 g	

⁴⁶⁸ most destabilizing environment for the stiffness-controlled actuator³. To separate the effects of model non-idealities, the red blocks in Fig. 10 are introduced
⁴⁷⁰ sequentially in the analysis.



Figure 10: Block diagram of the numerical model used to validate the theoretical analysis of coupled stability. The blocks describing the plant non-idealities are shown in red, and introduced sequentially in the numerical analysis.

 $^{^{3}}$ The choice of a particular form of perturbation (e.g. load force perturbation, load velocity perturbation, desired load position) does not change the coupled stability properties of linear systems, since the poles of the closed-loop transfer function are independent of the selected form of perturbation

5.1.1. Effect of plant non-idealities through numerical simulations



5.1.2. Experimental validation in the MR-SEA

The optimized control action was implemented in the MR-SEA, using values of desired stiffness K_{des} 25% and 50% higher than the physical stiffness of the spring. The p and z gains from Table 2 corresponding to the lowest-possible

⁴Although a higher range of stable stiffness regulation can be achieved by increasing the value of the force feedback control gain $k_{p,f}$, this case is not considered because the prototype is unstable for higher values of $k_{p,f}$, when $K_{des} < k_s$, due to unmodeled higher-order dynamics

V	p	z_{best}	$m_{e,MIN}$	$m_{e,MIN}^{sat}$	$m_{e,MIN}^{freq}$	$\left \begin{array}{c} m_{e,MIN}^{q} \end{array} \right $
κ_{des}	[1/s]	[1/s]	[kg]	[kg]	[kg]	[kg]
ine	500	150	0.12	0.23	0.54	0.54
$2k_s$	250	100	0.23	0.34	0.66	0.67
	100	50	0.58	0.60	0.93	0.94
ine	500	160	0.10	0.17	0.43	0.43
$1.75k_{s}$	250	105	0.19	0.26	0.53	0.53
	100	52	0.48	0.48	0.75	0.76
ine	500	180	0.07	0.12	0.31	0.31
$1.5k_s$	250	115	0.15	0.19	0.40	0.40
	100	55	0.37	0.37	0.57	0.58
ine	500	220	0.05	0.07	0.20	0.20
$1.25k_{s}$	250	133	0.10	0.11	0.25	0.25
	100	62	0.26	0.26	0.40	0.40

Table 2: Numerical model results - Minimum environment mass for coupled stability

value of interaction mass for coupled stability, in the worst case, are chosen (i.e. p = 500 for both cases and z = 220, 180 for $K_{des} = 1.25k_s$ and $K_{des} = 1.5k_s$, respectively). The stability of the system for these gains is proved in the most destabilizing case, i.e. the application of a constant force to the slider, with instantaneous load removal. In this case, the expected behavior is that of a sustained oscillation of the resulting spring-mass system (note that the stiffness compensator makes no attempt at reducing the load mass), but the energy dissipated through the control action (introduced in order to have the desired stability margins) provides the asymptotic convergence of the slider position - see Fig. 11(a). During high-stiffness display, the MR-SEA was commanded a reference position that was either constant $x_{L,des} = 0$ or oscillating $x_L(t) = A \sin(2\pi f_0 t) - A = 10 \text{ mm}, f_0 = 0.5 \text{ Hz}, \text{Fig. 11(b)}$. During the experiment, the subject alternated between the application of a continuous, roughly sinusoidal displacement

to the slider and the application of impulsive, impact-like displacement⁵. The stiffness transfer function $K_v(f)$ was estimated using non-parametric system identification, via the Welch method, using the following relation:

$$\hat{K}_{v}(f) = \frac{P_{yu}(f)}{P_{uu}(f)},$$
(17)

- with $P_{yu}(f)$ the cross-spectral density between the input (u) and output (y)and $P_{uu}(f)$ is the auto-spectral density of the input. To estimate the stiffness
- transfer function, the load position error $x_{L,des} x_L$ was selected as input, and the spring force $F_s = k_s(x_L - x_M)$ was selected as output. The estimated transfer function was resampled in a 51-elements, logarithmically spaced frequency
- ⁴⁹⁸ fer function was resampled in a 51-elements, logarithmically spaced frequency vector ([0.01 - 100] Hz), and normalized dividing by K_{des} , obtaining $\overline{K}_v(f)$.

Values of coherence higher than 0.8 were obtained up to approximately 6-7 Hz for both cases of $x_{L,des}$, and a Bode plot of the estimated normalized transfer function is shown in Fig. 11(c).

⁵⁰² Tulletion is shown in Fig. 11(c).

In the low-frequency range the system behaves as a pure spring, with elastic constant matching the desired value. The error in virtual stiffness approaches is higher than 3 dB up to a frequency that corresponds to 5-6 Hz, beyond which low coherence values are obtained and the transfer function estimate is no longer reliable. The phase of the estimated transfer function is negative, confirming

the non-passivity of the system when $K_{des} > k_s$.

6. Discussion and conclusions

This paper investigates the range of virtual stiffness values that can be stably and accurately displayed by a stiffness-controlled series elastic actuator (SEA),

⁵¹² in which the motor is a non backdrivable velocity source. As previously shown in the literature [14], a velocity-sourced SEA can be passively stiffness-controlled

to display a virtual stiffness K_{des} , only if $K_{des} \leq k_s$, with k_s being the physical stiffness of the compliant element. In our paper, we start by generalizing this

 $^{{}^{5}}$ This was done in the attempt to excite the system also at frequencies higher than the ones possible through manual perturbation at high stiffness



Figure 11: Experimental validation of the controller in the MR-SEA. (a) Application of an increasing load on the stiffness-controlled MR-SEA ($K_{des} = 1.5k_s$) with instantaneous load removal, resulting in decaying oscillations of the slider. (b) Perturbation of the MR-SEA slider when commanded a sinusoidal position tracking ($K_{des} = 1.5k_s$). After the initial transient, the force error is below 1 N before the subject applies perturbation forces to the slider (after 10 s). The desired force is calculated as $-K_{des}x_L$, and is compared with the measured force, i.e. $k_s(x_L - x_M)$. (c) Bode plot of the estimated normalized stiffness transfer functions, for two values of virtual stiffness and for both position tracking cases.

- finding, demonstrating that, in the context of linear control theory and with availability of only position information, no outer-loop force control action can
- be applied on the velocity-sourced system to result in a passive system, when a stiffness value $K_{des} > k_s$ is desired.
- We then analyze the effects on coupled stability of a stiffness controller (commanded to impose $K_{des} > k_s$), resulting in a non-passive impedance transfer function. We show that a frequency-limited lead action in the outer-loop stiff-
- ness controller results in the best accuracy and coupled stability performance,
- provided that the system is interacting with a load with inertia higher than a threshold m^* . Through our analysis, we derive the analytical expression that

- provides the minimum environment mass for stable coupled interaction, when a generic stiffness K_{des} is desired, as a function of the physical stiffness in the
- actuator and of the controller gains. Our analysis demonstrates that, when $K_{des} > k_s$, the value of minimum environment inertia required for a stable in-
- teraction increases with K_{des} with the scaling factor dependent on the choice of the controller gains.
- ⁵³² Using a similar methodology, we address the effect of a non-ideal inner velocity loop on haptic display performance and stability. In particular, we inves-
- tigate how the minimum value of interaction mass resulting in coupled stability can be minimized through placement of the stiffness compensator zeros and
- poles, considering realistic limitations in the bandwidth of velocity control, for non-backdrivable motors such as those used in [28] for haptic display of MR-
- compatible robots. We showed that with knowledge of the frequency limitations of the velocity inner loop (in our case knowledge of the time constant τ of a first-
- ⁵⁴⁰ order low pass filter), it is possible to further optimize the stability properties, by significantly reducing the value of minimum mass for a stable interaction, when
- the pole or zero at higher frequency is one decade below the cut-off frequency of the inner velocity loop.
- ⁵⁴⁴ We finally validate our model through numerical simulations and experiments in the MR-SEA II, a compliant actuator that includes a non-backdrivable
- ⁵⁴⁶ piezoceramic motor as a velocity source, demonstrating capabilities for stable rendering of virtual stiffness 50% higher than the physical spring stiffness.
- The main result of our analysis is that coupled stability of a stiffnesscontrolled compliant actuator can be obtained when the actuator is controlled
- to display a spring with elastic constant K_{des} , with $K_{des} > k_s$, provided that the minimum value of mass connected to the series elastic element is higher than a
- threshold, even in absence of friction. This finding is in contrast with an implicit assumption that has guided the development of most compliant actuators for
- ⁵⁵⁴ human interaction developed so far. Indeed, in most cases [14, 37, 38, 32, 39], the series elastic element has been designed to be substantially stiff, with the
- ⁵⁵⁶ idea of then controlling the actuator to display a lower virtual stiffness. The

choice of "stiff" springs for SEAs poses challenges on the selection of sensors used to infer interaction force, since they are usually required to have very high

resolution, in order to minimize quantization errors in the measurement of interaction forces. Consequently, some of the anticipated benefits deriving from 560

- elimination of a traditional force sensor, in terms of reduction in costs and size,
- can be lost in the attempt of incorporating the sensor(s) required to measure 562 the deflection of the compliant elements.
- The results of the presented analysis reveal opportunities for alternative de-564 signs of compliant actuators. Given a range of virtual stiffness values that need
- to be implemented through control, it is not necessary to design the compliant 566 element so that is stiffer than the stiffest-possible virtual environment for a given
- application. Instead, the results of this analysis show that the coupled system 568 will be stable if the system is designed so that it includes a minimum amount of
- inertia that is connected to the compliant element in all operating conditions. 570 Limitations to this approach can be provided by backlash, or in general of non
- co-located dynamics, between the series elastic element and the load. In the 572 presence of backlash, the value of load mass instantaneously decreases, poten-
- tially to a value that is lower than the threshold for coupled stability, until the 574 control action responds. The energy introduced in the system during impacts,
- coupled with delays in the position measurement and with velocity saturation 576 of the plant, is a major source of instability in the experimental implementation of SEAs. 578

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